

Overview of current processes and future developments in composites certification

A wind turbine industry perspective

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WHEN TRUST MATTERS



About DNV

About the Presenter

The Type Certification Process

The Present:

Safe Life – Load and Resistance Factor Design

Reliability

The Future:

Damage Tolerance



A global assurance and risk management company

159

years

~13,000 ~100,000

employees

customers

100 +countries



Ship and offshore classification and advisory



Energy advisory, certification, verification, inspection and monitoring



Software, cyber security, platforms and digital solutions

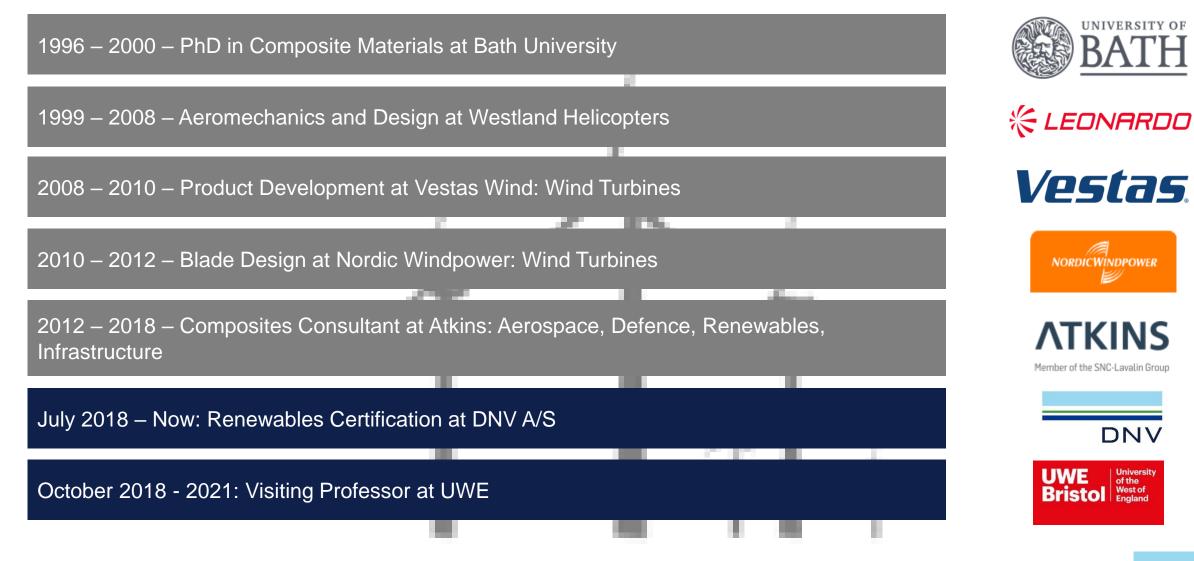


Management system certification, supply chain and product assurance



3

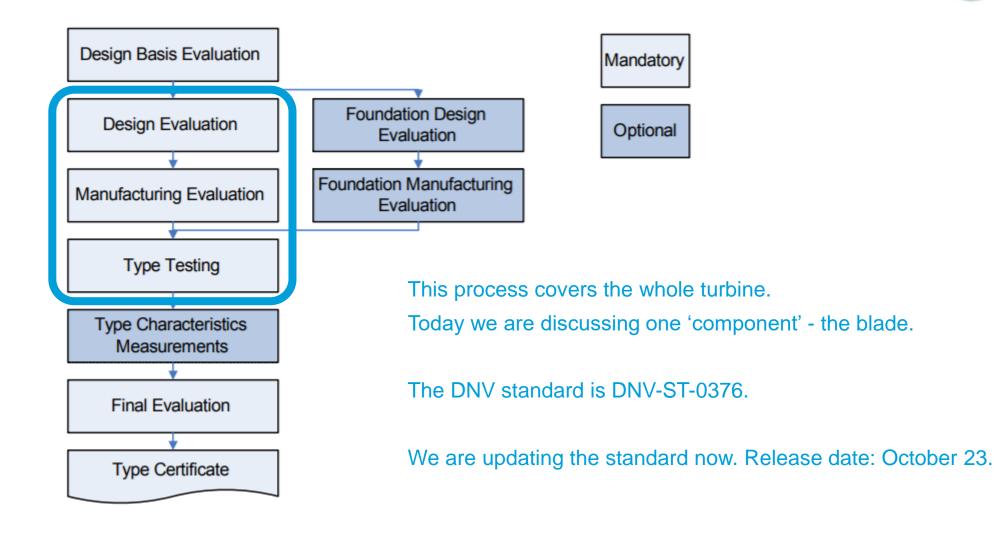
About the Presenter



Accredited by:



Type Certification Process



Present Requirements

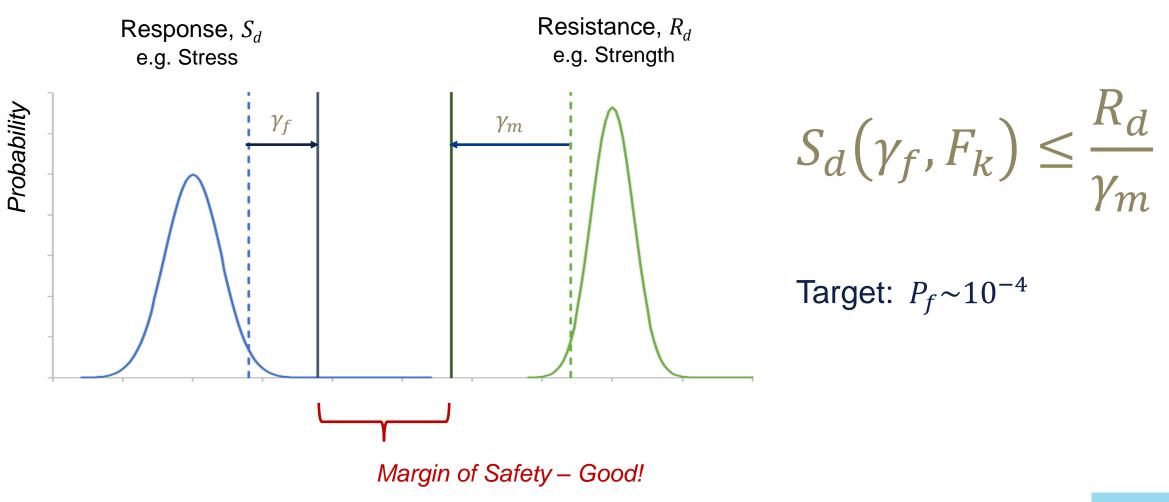


Design Evaluation – Mandatory Failure Modes

- Blade-tower strike (SLS)
- Laminate fibre failure (ULS and FLS)
- Buckling instability (ULS)
- Bonded joint failure (ULS and FLS)
- Inter-fibre failure (SLS)



Design Evaluation - Load and Resistance Factor Approach



Design Evaluation – Partial Material Safety Factors

 $\gamma_m = \gamma_{m0} \times \gamma_{mc} \times \gamma_{m1} \times \gamma_{m2} \times \gamma_{m3} \times \gamma_{m4} \times \gamma_{m5}$

S _d	\leq	R_d	
		γ_m	

 γ_m – the safety factor.

 R_d – the design value (e.g. material strength from coupon testing)

 S_d - the structural response under factored design loads (e.g. material stress or strain)

γ _{m0}	Base factor applied to analyses
γ _{mc}	Criticality of failure mode
γ _{m1}	Irreversible long-term degradation
γ _{m2}	Reversible temperature effects
γ _{m3}	Manufacturing effects
γ_{m4}	Accuracy of analysis methods
γ _{m5}	Accuracy of load assumptions
γ _m	Total

Manufacturing Effects

		Min.	Max.
Base factor	γ_{m0}	1.20	1.20
Criticality of failure mode		1.08	1.08
Environmental degradation		1.10	1.20
Reversable temperature effects	γ_{m2}	1.00	1.00
Manufacturing effects	γ _{m3}	1.00	1.30
Accuracy of analysis methods	γ_{m4}	1.00	1.25
Accuracy of load assumptions	γ_{m5a}	1.00	1.30
	γ_{m5b}	1.00	1.20
Total	γ _m	1.43	3.94
		(1.35)	

...which could be

- Variation in fabric placement
- Variation in fibre angle
- Wrinkles

this factor accounts

for uncertainty in

manufacturing

variation...

- Air entrapment in laminates
- Variation in fibre/resin ratio
- Contamination in resin
- Variation in mix ratio of resin
- Variation in mix ratio of glue
- Variation in degree of cure
- Contamination in bonding
- Surface preparation
- Variation in bond thickness
- Variation in bond width
- Variation of moisture in core
- Variation in cure shrinkage
- Variation in thermal shrinkage

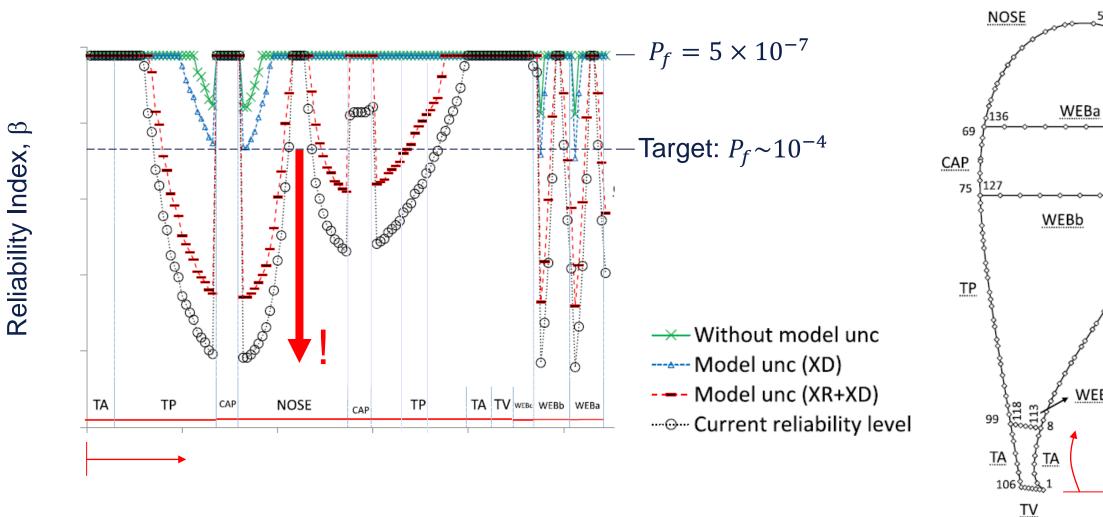
DNV

Design Evaluation – "Semi" Probabilistic Design Approach

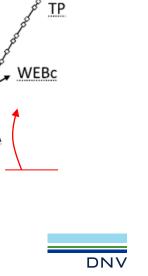
		Min.	Max.
Base factor	γ_{m0}	1.20	1.20
Criticality of failure mode	γ_{mc}	1.08	1.08
Environmental degradation	γ_{m1}	1.10	1.20
Reversable temperature effects	γ_{m2}	1.00	1.00
Manufacturing effects	γ _{m3}	1.00	1.30
Accuracy of analysis methods	γ_{m4}	1.00	1.25
Accuracy of load assumptions	γ _{m5a}	1.00	1.30
	γ _{m5b}	1.00	1.20
Total	γ _m	1.43	3.94

These factors are based on engineering judgement and experience. They are uncalibrated.

Reliability



'Effect of uncertainty sources on the reliability level of wind turbine rotor blades', Konstantinos Bacharoudis, Wind Energy 2018 : 21 : 1029-1045



53

128

119§34

40

CAP



Not Revolution!

1981

V15 (Vestas) 15m diameter 55kW Units sold = 975 MOD-2 (NASA/Boeing) 91m diameter 2MW Units sold = 3

But today wind turbines are really, really big.

RANZ BRACHT

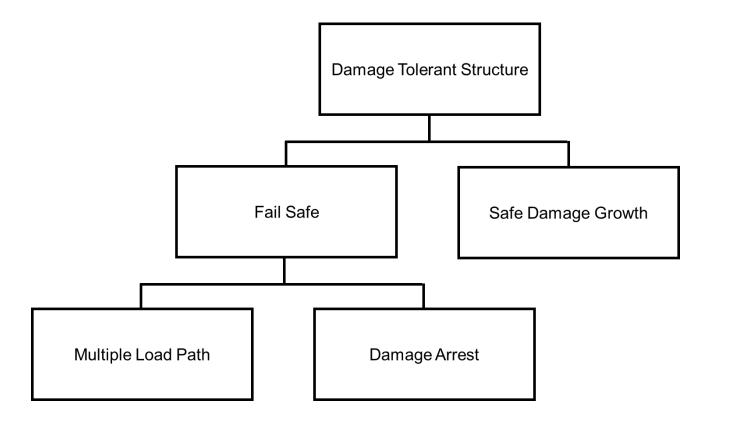
the consequence of defects and damage are higher and so we need to modernise certification to stay relevant to todays industry.

Future Requirements



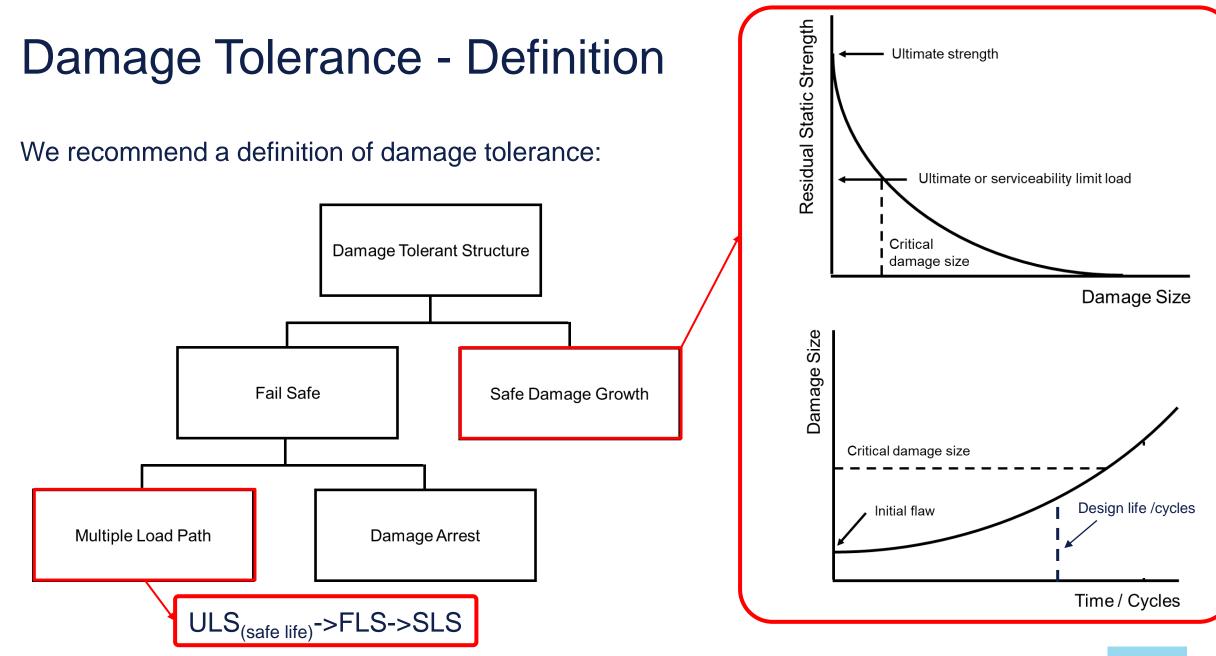
Damage Tolerance - Definition

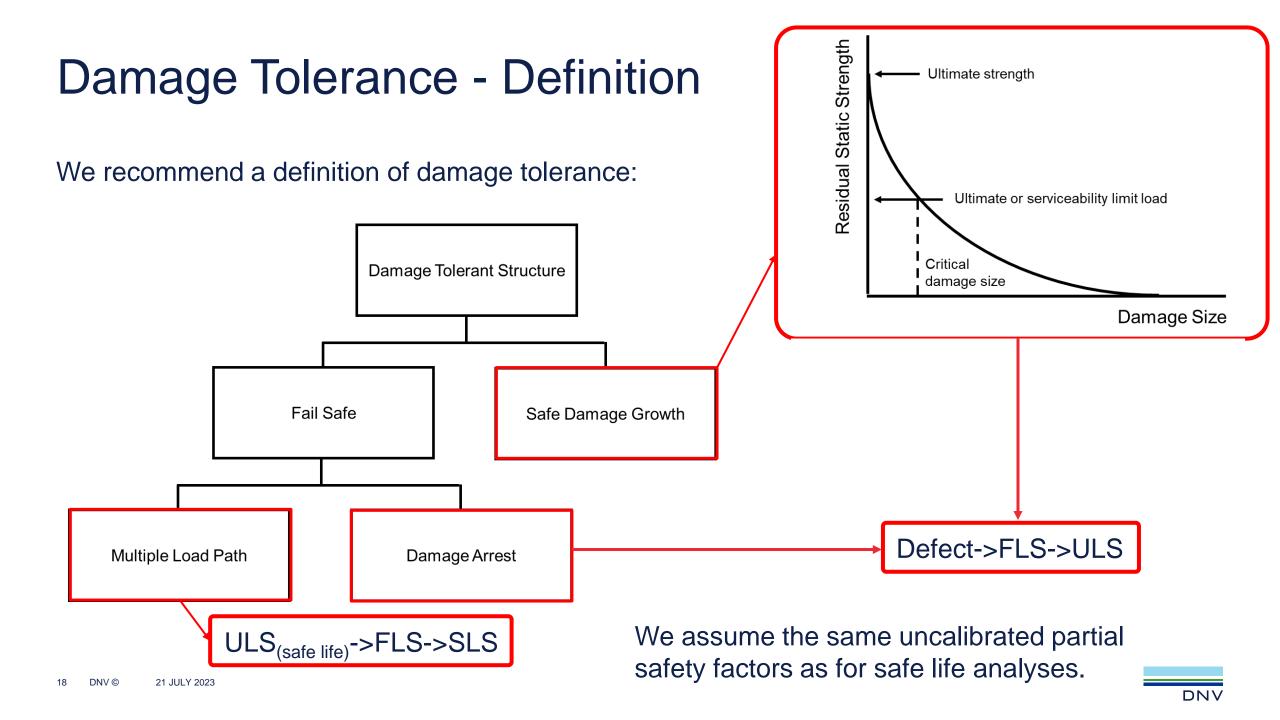
We recommend a definition of damage tolerance:



'Safe-Life and Damage-Tolerant Design Approaches for Helicopter Structures', Harold K. Reddick Jr., Applied Technology Laboratory, Army Research and Technology Laboratories (AVRADCOM)

'Damage Tolerance Evaluation of Fiber Reinforced Composite Tail Rotor Blades', Elif Ahci, 30th European Rotorcraft Forum





Damage Tolerance – Accepted Methods of Assessment

Damage Type	Numerical Analysis	Sub-Structure Testing	Note	
Fibre Failure		Acceptable		
Inter-fibre Failure		Acceptable	Numerical analyses = fracture mechanics	
Interlaminar Failure / Delamination	Acceptable	Acceptable	fracture mechanics	
Adhesive Failure	Acceptable	Acceptable	Sub-structure test = define	
Core Failure		Acceptable	characteristic values	
Facesheet Debonding	Acceptable	Acceptable		

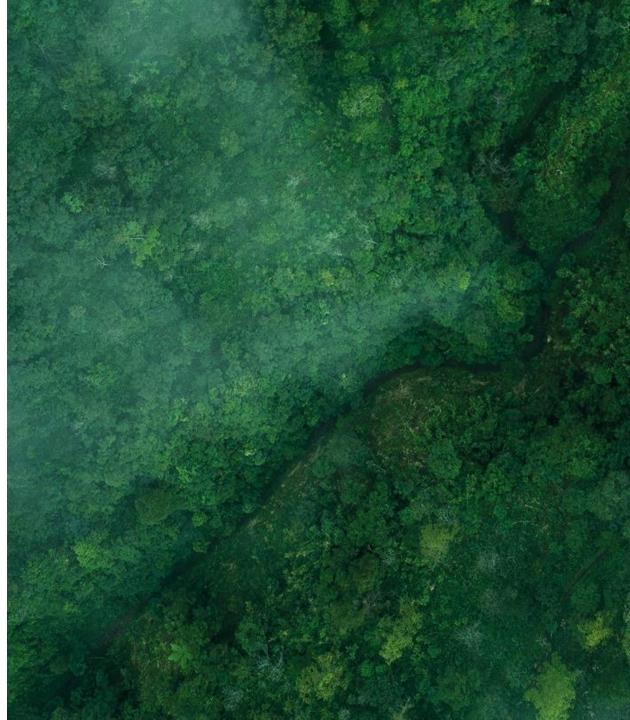
2 takeaways....



2 takeaways.....

- 1. The current partial material safety factors are based on engineering judgement and experience. They are uncalibrated, and work has shown that they are insufficient in producing the required target reliability.
- 2. We will use the same uncalibrated partial material factors for new damage tolerance assessment.

We could be happy for collaborative projects to better define these factors for future updates to the standard, DNV-ST-0376.



Any questions?

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